

Department of Mechanical and Aerospace Engineering

The Microbe Mixers

Group 243D

Michaela Baughn, Hugh Brittenham, Kyle Cunningham, Samuel Falzone, Ricardo Martinez, Darryl Mijares, Thierry Momplaisir, Asher Siddiqui

OWERING THE NEW ENGINEER TO TRANSFORM THE FUTURE



Meet the Team



Michaela Baughn



Hugh Brittenham



Kyle Cunningham



Samuel Falzone



Ricardo Martinez



Darryl Mijares



Thierry Momplaisir



Asher Siddiqui



DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Presentation Agenda

Hedgehog Concept Key Product Specifications Subassembly Highlights & Features

Engineering Analysis Product Testing Cost Summary

Hedgehog Concept

We aimed to improve our final product by utilizing our members' specializations in CAD, coding, and circuitry.

OUR PASSION

- Hands on manufacturing experience.
- The development of a robust code experience.

Function, Motion, Affordability

Combining different engineering disciplines to develop a highly functional product.

> OUR TALENT

Maintaining an optimally functional system with a low and reasonable price.

MARKET DEMAND

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Key Product Specifications

- Core XY Movement
- Usable Size:
 - 11.90" x 7.60" x 7.50"
- Lightweight Design:
 - 11.75 pounds
- Max Orbital Diameter:
 - 53.7 mm
- Accurate OD/FI Sensors:
 - Within 3% of calibration
- Heat Rated Parts:
 - Aluminum and PETG

Customer Need	Metric	Microbe Mixer's Implementation
1	Dimensions: 14" (356 mm) x 9.75" (248 mm) x 7.5" (191 mm)	11.90" x 7.60" x 7.50"
7	Assembled weight and packaging < 49 pounds	11.75 pounds
23	IP rating > IP-X5	IP – X5 Silicon Coating
25/26/27	Adjustable and variably linear, orbital and double orbital patterns at 25 mm diameter and 1200 rpm speed	Max Diameter: 53.7 mm Nema-17 Stepper Motors Used
28	Number of conical 15 mL tubes held ≥ 6 Number of conical 50 mL tubes held ≥ 6	# of 15 mL Clear Tubes = 6 # of 50 mL Clear Tubes = 6
33	Number of shut-off buttons ≥ 1	# of shut off buttons = 1
34	Number of pause buttons ≥1	# of pause buttons = 1
36	Max rated speed for 2 weeks at 70 C	Aluminum and PETG Parts
38	355nm wavelength light source Light source ≥ 6V	365nm UV LEDs
48	OD sensor reading is within 15% of fluid range calibration	Sensor Reading $\leq 3\%$

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Full System Design

Core XY movement

- Base Plate Subassembly
- Shaker Plate Subassembly
- UI Subassembly



UF

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Base Plate Subassembly



POWERING THE NEW ENGINEER TO TRANSFORM THE FUTURE

Herbert Wertheim College of Engineering

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Base Plate: Design Highlights



DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Base Plate: Design Highlights



UF

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Shaker Plate Subassembly



DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Shaker Plate: Design Highlights



DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Shaker Plate: Design Highlights



Lower Idler Positioner



New Design

 The chamfer and protrusion were eliminated to make a simpler design. The fastening location was moved to the top for ease of assemblance.



 Lower idler mount fastening hole was eliminated, while limit switch holes were raised for touching





DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

User Interface Subassembly



DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

User Interface: Design Highlights

- Redesigned for user safety
- Potentiometers tune the speed and path diameter
- Air vents relocated to the top
- Back panel designed to keep the electronics better protected
- LCD display shows easy to follow instructions and updates





DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

User Interface: Electronics



DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

OD/FI Subassembly

- Featured here is:
 - 15ml test tube
 - 30 ml test tube
 - Arduino Uno
 - 4 LDRs
 - 2 365nm UV diodes
- All housed within a 3D printed unit that will use a USB cord to connect this unit to the main UI
- Housing features a cover to protect electronics from any spillage, as well as keeps all components in one neat package.





UF

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Evolution of Design: OD/FI



OD/FI Analysis



18

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Material Ashby Plot Analysis



DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Shaker Top Plate Bending Analysis



$$\frac{1}{12}bh^3 = \frac{1}{12}(149.2 \text{ mm})(6.35 \text{ mm})^3 = 3.184 \text{ x} 10^3 \text{ mm}^4$$

 $\delta = -\frac{PL^3}{3EI} = -\frac{(11.1 N)(103.25 mm)^3}{3(68.9 x \, 10^3 MPa)(3.184 x \, 10^3 mm^4)} = 0.01856 mm$

POWERING THE NEW ENGINEER TO TRANSFORM THE FUTURE

Herbert Wertheim College of Engineering

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Drop Test Analysis

$$v_{f} = \sqrt{2gh} = 3.836 \frac{m}{s}$$

$$m = \rho_{PETG} \Psi_{enclosure} = 0.698 kg$$

$$c = \sqrt{\frac{E}{\rho}} = 1283.744 \frac{m}{s}$$

$$m = \frac{1}{283.744} \frac{m}{s}$$

$$F = \frac{\Delta p}{\Delta t} = \frac{mv_f - mv_i}{\Delta t} = \frac{2mv}{2\frac{L}{c}} = mv\frac{c}{L} = m\frac{c}{L}\sqrt{2gh} = 17805.88 N$$

$$\sigma = \frac{F}{A} = 712 \ kPa \qquad \qquad n_{FOS} = \frac{\sigma_t}{\sigma} = \frac{45.8 \ MPa}{0.712 \ MPa} = 63.62$$

POWERING THE NEW ENGINEER TO TRANSFORM THE FUTURE

Herbert Wertheim College of Engineering

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Vibration Analysis

 $z_r(t) = esin(\omega t)$

 $\ddot{z_r}(t) = -\omega^2 esin(\omega t)$

 $m\ddot{z} + c\dot{z} + kz = m_p e\omega^2 \sin(\omega t)$

$$r = \frac{\omega}{\omega_n} = 0.5$$



$$Z = \frac{m_p e}{m} * \frac{r^2}{\sqrt{(1 - r^2)^2 + (2\zeta r)^2}} = 1.55 \qquad \theta = \tan^{-1}\left(\frac{2\zeta r}{1 - r^2}\right) = 1.8$$

 $z(t) = Zsin(\omega t - \theta) = 1.392 mm$

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Orbital Diameter Calculation



Center of Rail Block in Relation to Top Edge of Base Plate

 $L_{max} = 150.66 mm$ $L_{min} = 96.96 mm$

 $d_{max} = L_{max} - L_{min} = 53.7 mm$

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Speed Calculation

$$F_f = W_{sys} * \mu = 150.7 \text{ } oz * (0.3)$$

= 45.21 oz

$$T = F_f * r_p = (45.21 \ oz)(0.54 \ in)$$

= 24.20 in - oz

$$780\frac{rev}{min} * \frac{2\pi}{1 \, rev} * \frac{1 \, min}{60 \, s} = 81.68\frac{rad}{s}$$

$$V_{belt} = C_p * \omega_{motor} = \left(81.68 \frac{rad}{s}\right) *$$
$$(0.085 m) = 6.98 m/s$$

$$\omega_{orbit} = \frac{V_{belt}}{C_{orbit}} = \frac{6.98 \frac{m}{s}}{0.0785 m} * \frac{1 rev}{2\pi rad} * \frac{60 s}{1 min} = 849.09 rpm$$





POWERING THE NEW ENGINEER TO TRANSFORM THE FUTURE

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING



DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Product Testing: IP-X5



As a preliminary test, the utilized IP-X5 Silicon spray was applied to the moisture indicators.



The strips were stored inside the table's housing and a constant water stream for the IP-X5 test was placed upon the design.



Only one indicator experienced failure and it was due to improper fastening. The tape holding the strip peeled the adhesive and altered the performance.

Product Testing: OD/FI

- 5 vials of florescent quinine were prepared and tested. Obtaining the concentration of quinine in the solution within a 15% range for all five was considered passing. (Florescent Intensity)
- 5 vials of a milk solution were prepared and tested. Obtaining the concentration of milk in the solution within a 15% range for all five was considered passing. (Optical Density)
- Successfully identified both milk concentrations and quinine concentrations within a range of 0 to 9% offset.



Future Testing

- For acoustic testing, the table must run at the full 350 rpm for 5 minutes while not exceeding a 50dBA sound level. Both linear and orbital patterns will be tested.
- For temperature testing, the table will be heated to a temperature of 70°C and must run at 350 rpm for 4 minutes. The same conditions must be met at a temperature of 4°C.
- The table must be fully functional after being dropped from a height of 75 cm onto concrete.

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Cost of Materials

PROTOTYPE

Part Type	Cost
OTS	\$ 843.45
Raw Material	\$ 84.48
Custom	\$ 20.99
Total:	\$ 948.92

MASS PRODUCTION

Part Type	Cost
OTS Parts	\$ 676.01
Raw Material	\$ 82.73
Custom	\$ 15.33
Total:	\$ 775.81

• OTS Parts

- Fasteners, pulleys, rails, electronics
- McMaster-Carr, Amazon, Digi-Key
- Raw Material
 - 6061-Aluminum
 - Base Plate, Shaker Plate
- Custom Parts
 - 3D-Prints (PETG)
 - Wall Enclosures, Tube Rack

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Cost of Assembly

Subassembly	Assembly Time (sec)
Base	615
Shaker Plate	490
User Interface	768
Total:	1873 sec = 0.52 hours

The average salary for an assembler in Florida: \$16.27/hour

Total Assembly Cost: \$8.46



Cost of Manufacturing

Subassembly	Assembly Time (Hrs.)
Base	3.5
Shaker Plate	2.5
User Interface	0
Total:	6 hours

The average salary for a machinist in Florida: \$22.52/hour

Total Assembly Cost: \$135.12

POWERING THE NEW ENGINEER TO TRANSFORM THE FUTURE

Herbert Wertheim College of Engineering

DEPARTMENT OF MECHANICAL AND AEROSPACE ENGINEERING

Total Cost

Prototype







Design Summary

- Expansions in the design make for a multifaceted project.
- Room for improvement in most aspects of the project.
- Low cost and high versatility show enough promise that the concept should be completed.





Department of Mechanical and Aerospace Engineering

Thank You. Questions?

POWERING THE NEW ENGINEER TO TRANSFORM THE FUTURE